

Back-Projecting Secondary Craters Using a Cone of Uncertainty

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Abstract

In this paper we present an extension to the Large Crater Clustering (LCC) tool set which places a cone of uncertainty around the trajectories of secondary impact craters to determine potential locations of source craters. The LCC tool set was a first step in the spatial quantification of primary and secondary cratering processes, which allows planetary geologists to accurately estimate the geologic age of a celestial surface. This work builds on the LCC tool set by accounting for the ambiguity of flight path trajectories through a Python script that leverages ArcGIS's ArcPy library. We chronicle the mechanics of the script, which creates geodetically correct cones then counts them within equally sized cells of a vector grid. We describe the process that was used to derive the shape of the cone and provide parameters for the sizes of the cones and the grid. We demonstrate that the cone of uncertainty has the ability to compensate for error in secondary crater trajectories by introducing deviation in the trajectory bearing and comparing the predicted primary crater location. We use two study areas on Mars as well as the entire

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lunar surface to illustrate the usefulness of the extension as an aid to human interpretation of back-projections.

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1. Introduction

Crater counting, crater size frequency distribution assessment, and the identification of the relationship between primary and secondary craters are tools used to determine the age of a celestial body (Shoemaker et al., 1963; Barlow, 1988; McEwen and Bierhaus, 2006; Robbins and Hynek, 2014, 2012; Platz et al., 2013). Shoemaker (1962) identified secondary craters, those resulting from ejecta being propelled from a primary impact to be deposited elsewhere on the surface, as one of the main issues of using crater size-frequency distributions for temporal analysis. These secondaries contaminate crater counts to bias the data in a given area toward older geologic dating. The distinction of primary versus secondary craters must be approximated in order to perform accurate surface aging, and allows an opportunity to further our understanding of cratering processes.

Primary and secondary craters have distinctive patterns, morphologies, sizes, and orientations that provide opportunities for traditional spatial analysis (McEwen and Bierhaus, 2006; Laura et al., 2017). The average impact angle of a primary impact crater is approximately 45° (Shoemaker, 1962; McEwen and Bierhaus, 2006) and results in secondary impact craters with

19 a range of impact angles and morphologies (McEwen and Bierhaus, 2006).
20 McEwen and Bierhaus (2006) suggest that secondary impacts are frequently
21 smaller (<5% of primary impact diameter), shallower, and more elliptical
22 than primary craters, appearing to have been scooped out from the sur-
23 face. Spatially, secondary craters can often be found in a tight ring around
24 the primary that formed them or clustered in rays emanating from the pri-
25 mary (Preblich et al., 2007; McEwen et al., 2003). The shape of secondary
26 clusters, catenae, and lineaments provide a means to estimate the primary
27 crater location by back-projecting a trajectory either through multiple fea-
28 tures (polylines) or through semi-major axes of bounding ellipses (clustered
29 point data or crater clusters digitized as bounding ellipses) and locating in-
30 tersections between these trajectory estimations. Back-projecting secondary
31 craters can be done through a combination of human interpretation (McEwen
32 et al., 2003) and geospatial analysis (Laura et al., 2017).

33 Skinner and Nava (2011) and Laura et al. (2017) identify the need for a
34 quantitative method to study primary and secondary cratering. The Large
35 Crater Clustering (LCC) tool set was developed to respond to this need. The
36 LCC tool set transforms user supplied, secondary craters into trajectories
37 which are back-projected along great circle arcs to estimate potential source
38 crater locations. The tool set represents a first step in quantitatively exam-
39 ining secondary to primary crater flight trajectory relationships and provides
40 a means to further studies that seek to empirically quantify and constrain
41 secondary ejecta flight properties. The LCC tool set is comprised of the fol-
42 lowing five tools (processing steps) to support primary location estimation:
43 (1) calculating the nearest-neighbor distances among a set of independently

44 digitized input crater features, (2) using these distances to identify clusters
45 of craters, (3) fitting an ellipse to bound the clusters, (4) back-projecting
46 the ellipse along the semi-major axis to produce a flight path trajectory, and
47 (5) intersecting these trajectories (polylines) to estimate the primary impact
48 (Laura et al., 2017). When features are digitized as polyline or bounding el-
49 lipses the tool can begin running at step four. In conjunction with the LCC
50 tool, Laura et al. (2017) implement a value-by-alpha visualization developed
51 by Roth et al. (2010a) that uniformly buffers secondary trajectories to some
52 user defined distance and utilizes the transparency (or alpha) component
53 of the buffered geometry to support visual identification of areas of highest
54 intensity, and therefore most likely primary impact location.

55 The LCC tool set has spatial analysis issues that can result in significant
56 ambiguity in quantitative primary impact crater identification. Laura et al.
57 (2017) identify issues regarding digitization, the impact of cluster outliers,
58 edge effects, and the Modifiable Areal Unit Problem (MAUP)¹ (O’Sullivan
59 and Unwin, 2010). In this work we directly address the error in the resulting
60 trajectories and comment on the MAUP in the search for primaries. This
61 paper describes our extension to the LCC tool set which places a cone of
62 uncertainty around the projected secondary flight trajectory to account for
63 ambiguity in ejecta flight paths and in the digitization process.

64 The balance of this paper is arranged as follows. In Section 2 we describe a
65 Python script that fits a cone of uncertainty to secondary crater trajectories.
66 In Section 3 we describe a simulation which introduces sources of trajectory

¹Bias in the analysis resulting from the placement of the study area boundary or the scale that was used.

67 error in a test area to evaluate the effectiveness of the cone. Section 4 explores
68 three case studies, one from the Moon and two from Mars, to evaluate the
69 cone of uncertainty in real scenarios. Finally, in Section 5, we comment on
70 the validity of the cone of uncertainty and look ahead to future work.

71 **2. Methods**

72 In this work, we address issues of ambiguity in back-projection through
73 the use of the cone of uncertainty and provide quantitative methods for
74 the computation and interpretation of trajectory intersections. Using the
75 back-projected secondary flight trajectories (from the LCC tool) we create
76 two outputs: (1) a cone of uncertainty fit around each trajectory and (2)
77 a vector grid whose values represent the number of overlapping cones. The
78 former seeks to account for spatial ambiguity and digitization bias and the
79 latter seeks to reduce the need for visual identification of high trajectory
80 intersection locations, i.e., qualitative interpretation. In this section, we
81 describe the selection of shapes the uncertainty buffer may take and the
82 implementation of the LCC extension.

83 The cone of uncertainty is perhaps most widely known in tropical storm
84 forecasting, where an event's location becomes increasingly uncertain with
85 time and distance. In this construct the cone serves as an aid to forecast the
86 future. Figure 1 shows the basic construction of a cone of uncertainty. When
87 looking at secondary craters we view the same phenomenon in reverse. In this
88 case the event has already occurred with a known endpoint but an unknown
89 starting point. The best fit flight path of ejecta has increasing uncertainty
90 as the distance from the secondary (toward the starting point) increases. By

91 looking at an individual crater feature, we can create a cone with increasing
 92 width away from the feature representing the greater uncertainty of its ori-
 93 gin. The uncertainty in secondary crater flight trajectories is known to exist
 94 with constraints developed by Popova et al. (2007), but there is currently
 95 no quantitative method that exists to incorporate this uncertainty into the
 96 back-projection process.

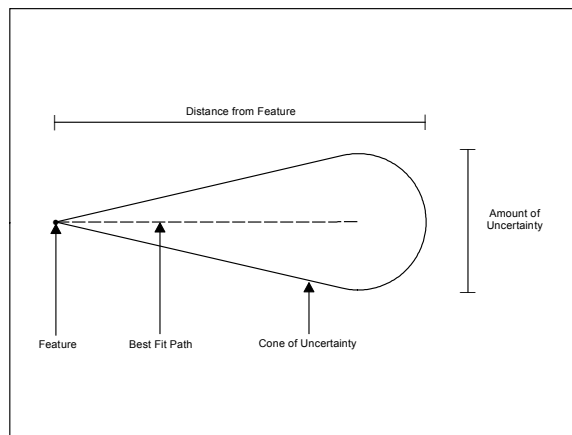


Figure 1: Cone of uncertainty shown as a shape that follows a best-fit trajectory. Width of the cone corresponds to the amount of uncertainty in the best-fit path, where a wider shape represents greater uncertainty.

97 2.1. Cone of Uncertainty Creation

98 We have extended the LCC tools by creating a Python (van Rossum and
 99 Drake, 2016) script which uses the output of the fourth processing step, a set
 100 of vector features representing the back-projection, to fit a cone of uncertainty
 101 around the trajectory and ultimately estimate the primary impact. The
 102 creation of the cone has four basic steps: (1) creating the trajectory within
 103 the LCC tool set, (2) buffering the endpoints of the trajectory, (3) creating a

104 convex hull to encapsulate the space, and (4) adjusting the shape of the hull
105 to account for the curvature of the surface. These four steps are depicted in
106 Figure 2.

107 Use of the LCC tool and the creation of the line trajectory is described
108 by Laura et al. (2017). The output of the LCC back-projection tool is a
109 bi-directional line, with each segment ending at the secondary crater. In
110 most cases, and without additional geomorphic knowledge of the feature, it
111 is unknown from which direction along the semi-major axis the ejecta flew
112 (McEwen and Bierhaus, 2006). The second phase of the cone creation, shown
113 in (b) of Figure 2, buffers the starting and ending points of the trajectory.
114 These buffers establish limits on the size of the cone where larger buffers
115 indicate larger cones and greater uncertainty. More specifically, the amount
116 of uncertainty is given by the size of the buffer around the starting point
117 of the trajectory, the furthest point from the secondary. The buffer is a
118 geodetic circle with a user defined radius. The user has the ability to enter
119 multiple buffer sizes. If multiple sizes are entered, a separate layer of cones
120 will be created for each entered size. The buffer for the end of the trajectory,
121 near the secondary, is a small standard circle that represents the minimal
122 amount of uncertainty in the region. The small buffer also prevents potential
123 overlap with the ending buffer of the other half of the bidirectional line for
124 that secondary.

125 The next step creates a polygonal shape to encapsulate the line endpoint
126 buffers. The two buffers are merged into one multipart polygon, then a
127 convex hull² envelops the space around it by wrapping around the outside

²The smallest shape that can be created that contains all given points where no line

128 half of each buffer and connecting in between them. The convex hull allows
 129 the amount of uncertainty to uniformly increase from the secondary to the
 130 buffer around the starting trajectory. This is depicted in (c) of Figure 2. The
 131 resulting shape is a planar hull that follows the projection being used but not
 132 the shape of the surface, therefore the distance to the trajectory is unequal
 133 along the entire feature and in some cases the trajectory may be outside the
 134 hull. The final step in the cone creation process corrects for this by altering
 135 the hull so that it represents the shortest distance along the surface and
 136 follows the path of the trajectory. The final output is shown as (d) in Figure
 137 2.

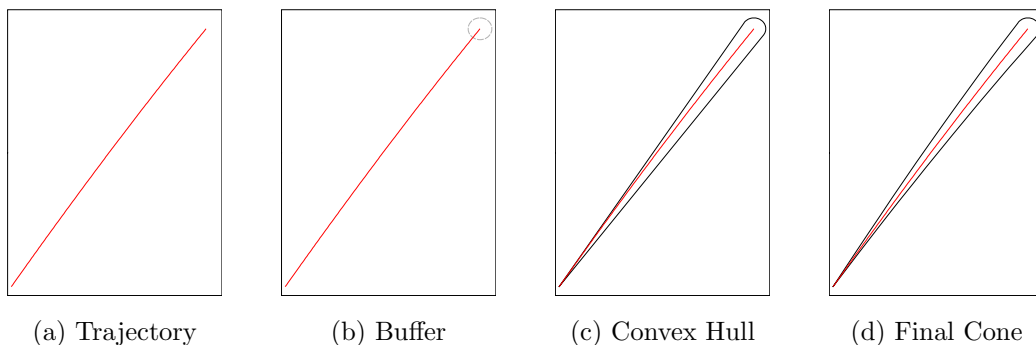


Figure 2: Four basic steps in building the cone of uncertainty are (a) the starting trajectory (b) buffering the starting point according to a user-defined distance (c) creating a convex hull of the shape (d) geodetically altering the hull to arrive at the final cone.

138 *2.2. Counting Intersecting Cones*

139 Once the cones have been created, the final section of the LCC exten-
 140 sion script focuses on identifying source craters by counting the number of

drawn between those points extends outside the shape.

141 cones that intersect with a uniform grid. This is a similar concept to the
142 Value-By-Alpha approach developed by Roth et al. (2010b) and described
143 for use in identifying primary craters by (Laura et al., 2017). While Laura
144 et al. (2017) sifted for primaries qualitatively, the grid approach taken in the
145 LCC extension aims to suggest source craters quantitatively. Our extension
146 supports exploration of issues of scale (MAUP) by allowing for multiple runs
147 with differing cell sizes to create a more complete picture of the crater rela-
148 tionships. The grid approach aids in human interpretation by allowing the
149 user to identify patterns within the study area. Examples of interpretation
150 of the data using this method are given in Section 4.

151 Counting the overlapping cones is a three step process using the LCC
152 extension: (1) creating a grid, (2) performing a spatial join between the
153 layer of cones and the grid, and (3) depicting the resulting number of counts.
154 These three steps are illustrated in Figure 3. The extension requires an initial
155 specification of a projection for subsequent processing. The data from the
156 LCC tool set should be re-projected to an equal-area projection to ensure
157 the grid cells are of equal size. Other projections may distort the grid sizes
158 and resulting cone counts.

159 The analysis grid is created by utilizing the fishnet function within the
160 ESRI ArcGIS ArcPy site package. The extent of the grid is set to match
161 that of the original trajectory layer. A user defined size, input at the begin-
162 ning of the script, determines the size of each grid cell. With the grid and
163 cone layers in place, a spatial join is performed that counts the number of
164 features in the cone layer that overlap with each feature (cell) in the grid.
165 A new layer is created, having identical spatial properties as the grid, which

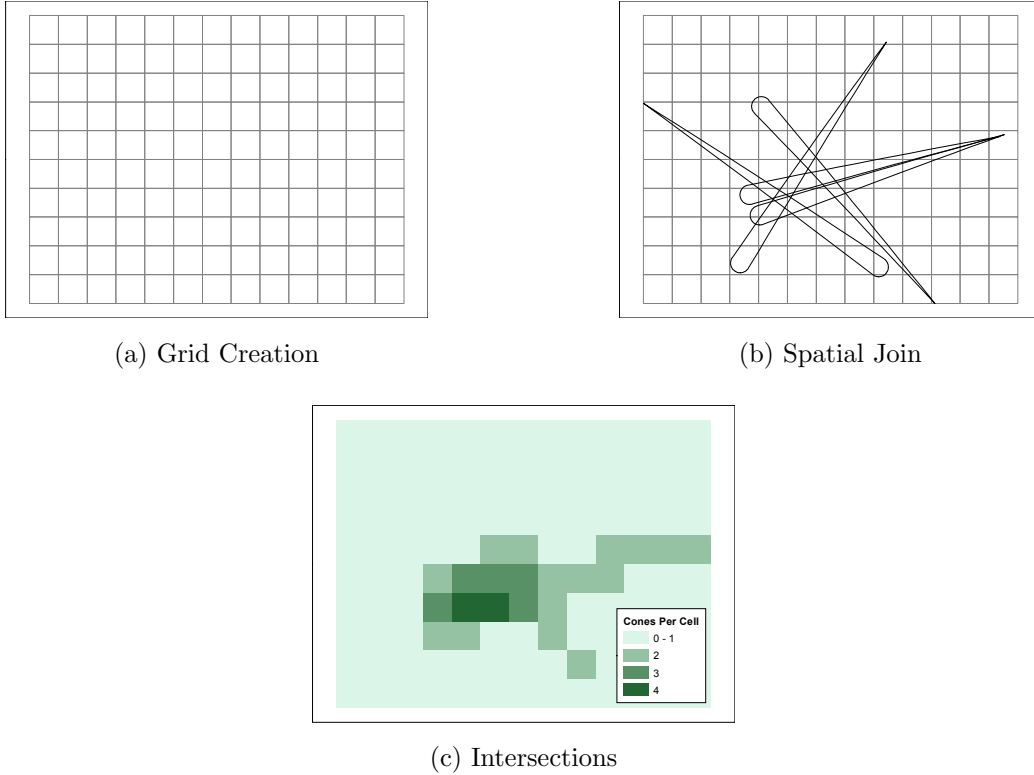


Figure 3: Quantitatively searching for a primary crater is a three step process: (a) creating a grid (b) intersecting it with the trajectory cones (c) thematically representing the overlap counts.

166 contains this count. In order to make use of this information, the data are
 167 exported out to an ArcMap .mxd document. The method of classification
 168 and thematic representation occur external to the LCC extension. Examples
 169 of these depictions are shown in Sections 2.3, 3, and 4.

170 *2.3. Testing and Parameterization*

171 We empirically tested five cone of uncertainty shapes with the goal of
 172 determining which would be the best predictor of a primary crater. These

173 shapes were designed to test which area(s) along the trajectory typically in-
174 tersect the source crater and how the size of the shape affected predictability.
175 The first round of testing was intended to compare the cone against the the-
176 oretically most predictive sections of the line trajectory. Figure 4 illustrates
177 each of the five tested shapes with the initial polyline depicted at the center.
178 These shapes are as follows:

- 179 1. A single uniform buffer, intended to show some variability in the line
180 trajectory that does not deviate with distance. This is the simplest fit
181 around a trajectory and was selected as a ‘proof of concept’ of using a
182 shape, in lieu of a line, to predict a primary. When compared to the
183 output of the cone test this geometry provides an idea of how strong
184 the middle of the cone predicts a primary relative to its periphery.
- 185 2. A circular buffer placed around the starting point (point opposite the
186 secondary), or beginning of the modeled flight path, where the primary
187 crater should be. This buffer is similar to the widest section of the cone
188 and allows us to see how accurate this portion of the shape is, and how
189 well the length of the LCC line can pinpoint the primary. We note that
190 that the LCC trajectory tool supports user defined trajectory length,
191 making estimation using this method a function of both the selected
192 buffer size and the user defined flight distance length.
- 193 3. A uniform cone, the classic shape from hurricane modeling, where the
194 girth becomes wider as the distance away from the secondary increases.
195 This was the most intuitive shape because of the uncertainty of the
196 trajectory between the secondary and primary. In theory this shape
197 should be the most predictive so if shapes (1) or (2) are significantly

- 198 more predictive the cone will need to be altered.
- 199 4. A cone given a weight corresponding to the ellipticity of the secondary
200 crater feature. The less predictive, circular crater clusters receive a cone
201 that has a lower weight. It was an attempt to see if the cone could be
202 improved by tying the overlap count to the ellipticity of the cones rather
203 than the number of cones. If successful, it should significantly improve
204 the prediction of Zunil.
 - 205 5. A nested cone which keeps the intuitive cone approach but gives higher
206 weight to the middle and end of the trajectory, the area near the trajec-
207 tory and closest to the secondary, which are the most predictive aspects
208 of the cone as determined from the first three shapes. In practice this
209 is not one cone but three on top of each other so a grid cell intersecting
210 the last half of the trajectory line would be given a count of three, not
211 one as in the case of the uniform cone. Thus it implicitly gives more
212 weight to those parts of the cone deemed most predictive.

213 Each shape was fit around trajectories created in the LCC tool set from
214 a sample of known secondaries originating from Zunil Crater, Mars. Figure
215 4 displays these basic shapes, their fit to the Zunil trajectories, and the
216 overlap counts for each of the tests. As explained in Section 2.2, the highest
217 intersection counts per cell suggest the location of a primary, in this case
218 Zunil, and are displayed with darker shading. The red circle is the actual
219 location of Zunil.

220 Of the three initial shapes, only the starting point circular buffer was
221 unable to suggest the general location of Zunil. It is assumed then that the
222 uncertainty near the beginning of the trajectory is relatively large. How-

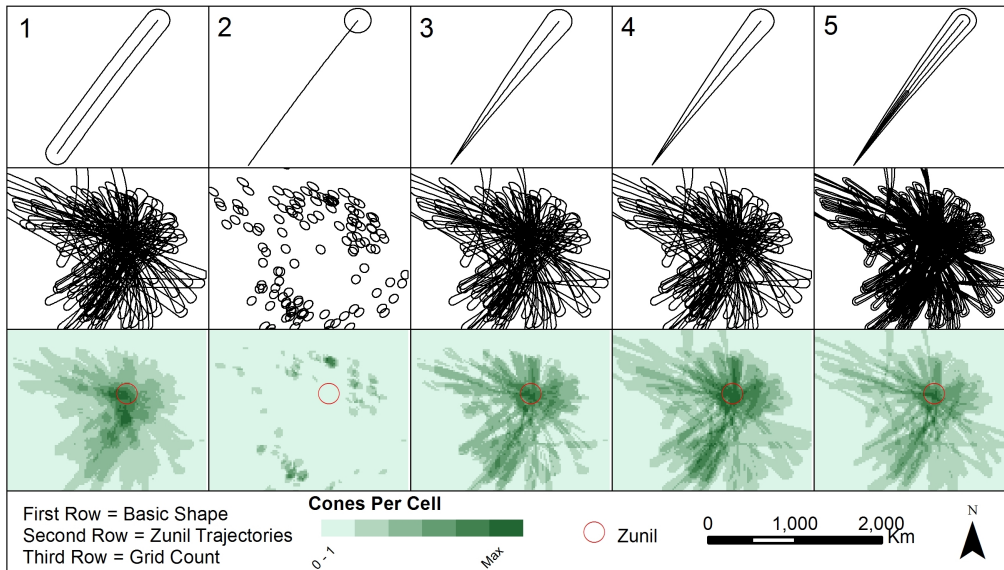


Figure 4: Basic cone shape, shape placed around Zunil secondary trajectories, and overlapping shape counts for (1) simple uniform buffer, (2) circular buffer around beginning of trajectory, (3) uniform cone, (4) weighted cone based on ellipticity, and (5) nested cone. Uniform cone is the best compromise of source crater predictability and script run time.

223 ever, it should be noted that other crater systems may align well with the
 224 beginning of the modeled flight path. Looking at the location of Zunil and
 225 the other shapes it is clear that Zunil is typically within the last half of the
 226 trajectory. The uniform buffer and cone both predicted the general location
 227 of Zunil. The cone was more precise in its prediction, and has a more intu-
 228 itive shape given the distance-uncertainty relationship described in Section
 229 2. Therefore we determine from these three initial tests that the cone can
 230 be used to predict the location of a source crater, and that the area closest
 231 to the trajectory and away from the starting point are the most predictive.
 232 With this knowledge, we tested two additional methods to see if the cone

233 could be improved.

234 The cone shape based on ellipticity was not significantly more predictive
235 of Zunil than the uniform cone, and required additional run time. Within
236 the LCC tool the user has the ability to place a threshold on the ellipticity of
237 secondary features when creating the trajectories, and therefore we suggest
238 including a cutoff here to include only those trajectories whose directional dis-
239 tribution can be confidently assumed. While it is the most complete picture
240 of the trajectory, the successes of the nested cones are offset by the increased
241 required storage and run time. Therefore, the uniform cone was selected as
242 the final shape, having the most desirable trade-off between predictability,
243 run time, and disk space.

244 The usefulness of the LCC extension is tied not only to the shape of
245 uncertainty being used, but the size that shape takes as well as the grid that
246 counts it. For the cone it is assumed that the uncertainty near the secondary
247 crater is low and thus the minimum width is essentially zero, as the location
248 of the secondary is known without modeled ambiguity. Popova et al. (2007)
249 suggest that the maximum width dispersion of a given set of secondaries from
250 a primary (e.g., a single trajectory) is at most 25 kilometers. We utilize this
251 constraint as a starting point in cone selection. While a known trajectory
252 may have a width dispersion of 25 kilometers, we must expand this value to
253 account for ambiguity in the digitization and trajectory estimation processes.
254 For the Zunil system, cones having starting trajectory buffer diameters of 5,
255 10, 25, 50, and 100 kilometers were tested. The resulting cones counted
256 over a grid with 400 square kilometer cells is shown in Figure 5. With no
257 significant difference in run time, larger cones tended to distill the overlaps

258 down to a useful level.

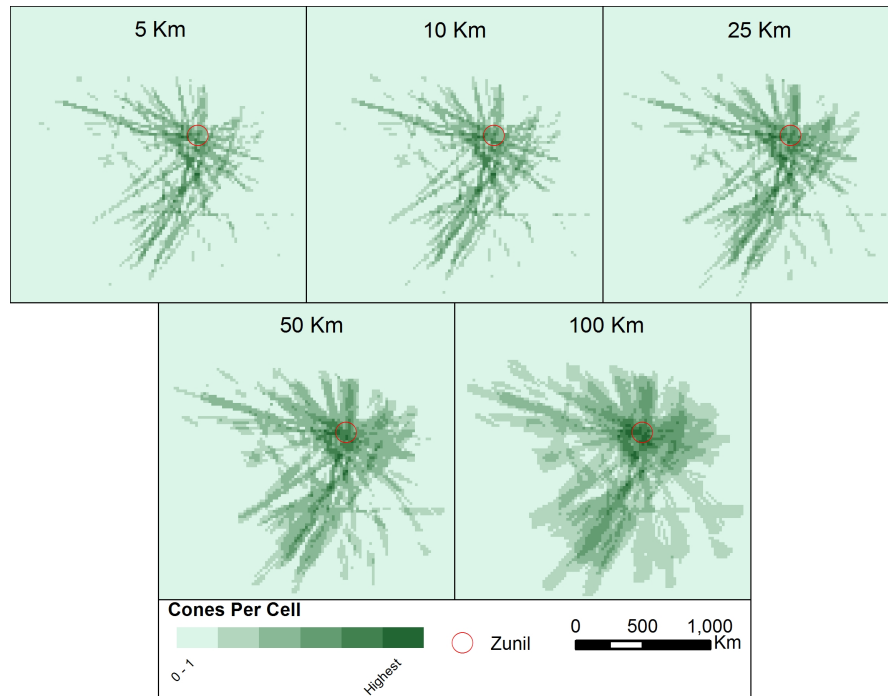


Figure 5: Cones of various sizes enumerated against a grid with 400 square kilometer cells in the Zunil test area. The largest cones create the most intersections and a more complete picture of the region.

259 The grid cell size is dependent on the spread of trajectories, the amount of
260 detail required by the user, and the amount of run time desired. To test the
261 size of the grid cells counting the cone intersections, cones having a maximum
262 width of 50 kilometers were intersected with grids having cell sizes of 25, 100,
263 400, 2,500, and 10,000 square kilometers for the Zunil test area. The results
264 of this test are shown in Figure 6. The two smallest grid cell sizes created
265 numerous areas without intersections and made visual pattern recognition
266 more difficult. The two largest grid cell sizes tended to mask much of the

267 useful detail for human interpretation. The grid with 400 square kilometer
268 cells provided the best compromise of detail and run time.

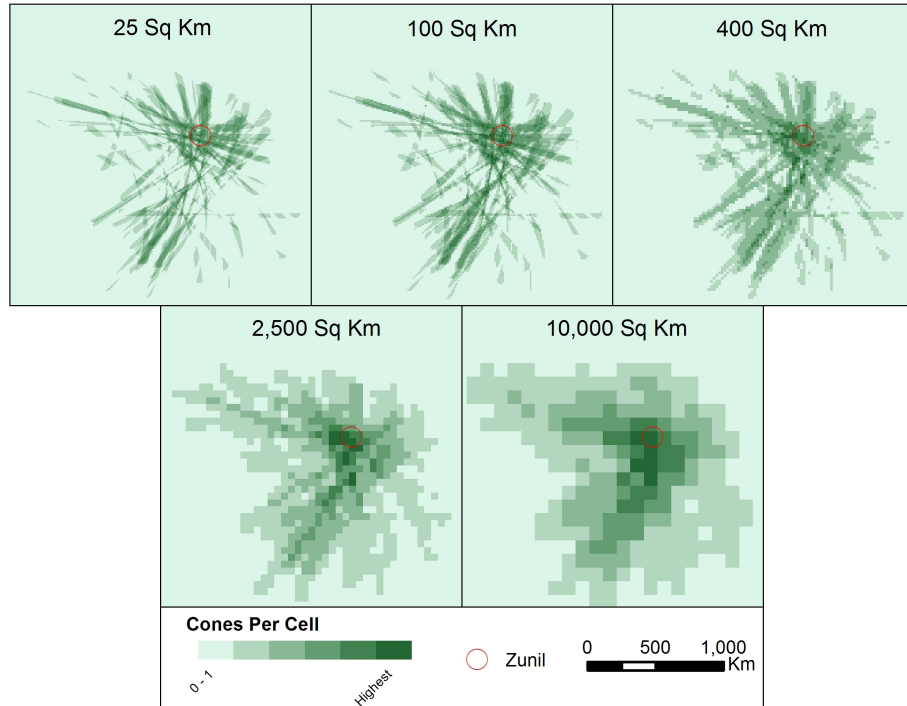


Figure 6: 50 kilometer cone of uncertainty enumerated against grids of varying cell sizes in the Zunil test area. Largest grid sizes run fastest, but with least detail.

269 Table 1 shows the increasing amount of time required for more granular
270 cells. It also shows the runtime for cones of various sizes run over a 400
271 square kilometer grid. The algorithm is more sensitive to the resolution of
272 the grid cell and the number of grids being run. The balance of run time
273 and detail required will depend on the user and system. The user has the
274 ability to run multiple cone and grid options simultaneously, and may choose
275 to do so to establish the most useful combination. It is recommended that
276 the user try several cone sizes and compare the output to gather the most

277 information possible for the area at hand. A suggested default for a grid size
 278 in a Martian quadrangle is 400 square kilometer cells.

Cone Size (Km)	Grid Size (Sq Km)	Runtime
50	10,000	x
50	400	1.3x
50	25	7.0x
50	25, 100, 400, 2,500, 10,000	9.3x
5, 10	400	1.1x
5, 10, 25	400	1.1x
5, 10, 25, 50	400	1.2x
5, 10, 25, 50, 100	400	3.4x

Table 1: Script runtime for Zunil secondaries with various cone and grid selections.

279 3. Accounting for Uncertainty

280 3.1. Sources of Error

281 The usability of the cone of uncertainty in back-projecting secondary
 282 craters is contingent on its effectiveness in identifying the location of source
 283 craters and compensating for error in the trajectory itself. A line back-
 284 projection is a best estimate of a dynamic process having uncertainty arising
 285 from the following factors:

- 286 • Inevitable uniqueness in the individual secondary crater trajectories
- 287 • Continually developing science in the flight path of impact ejecta

288 • Uncertainty in the digitization of crater features

289 Although the first source of error almost certainly will exist, it is difficult
290 to capture the deterministic randomness that any flight path might have.
291 There has been significant research to understand impact and ejection events,
292 and the resulting craters. Please see Vickery (1986, 1987), Popova et al.
293 (2007), Melosh (1984), and Schultz and Gault (1985) for the details of this
294 research. The error associated with the digitization is a combination of the
295 skill, knowledge, and attention to detail of the user and the spatial resolution
296 of the imagery from which the digitization is developed.

297 *3.2. Error Simulation*

298 We have run a series of Monte Carlo simulations that attempt to mimic
299 the error identified in Section 3.1. The Zunil secondary crater trajectories
300 were replicated 500 times, with each replication adjusting each trajectory
301 between -10° and $+10^\circ$ sampled from a uniform random distribution. The
302 trajectories themselves are shown in Figure 7 along with the simulated tra-
303 jectories. Cones having 50 kilometer diameters were fit to each trajectory
304 within these simulations and the overlaps were counted within a grid hav-
305 ing 400 square kilometer cells. For each simulation, the grid cell containing
306 the highest number of overlaps was identified then combined with the max-
307 imum cell from every other iteration. This was repeated with cones having
308 a diameter of 100 kilometers. If the cone is successful in compensating for
309 the introduced error, the maximum cell should be consistent throughout the
310 iterations and drift should be minimal.

311 The two images in Figure 8 show the cells that had the maximum number
312 of overlaps in at least one iteration, with the count representing the number

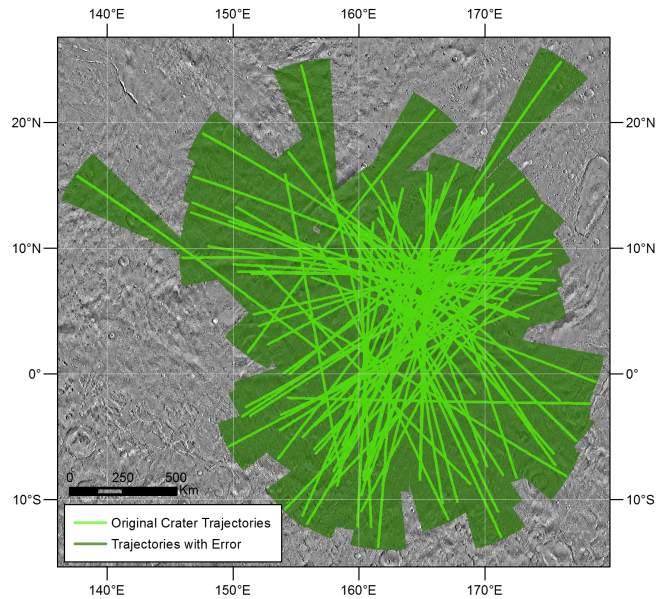


Figure 7: Original trajectories from Zunil with trajectories of 500 error-induced simulations.

313 of simulations where that cell was the max. Many iterations had multiple
 314 cells with the maximum count, so the totals are greater than 500. The red
 315 circle is the actual location of Zunil. In general the amount of drift in the
 316 maximum cells is minimal and the vast majority of iterations predict Zunil
 317 well. In both the 50 kilometer and 100 kilometer cone cases, the area of
 318 maximum overlap tends to be within a block of cells west of Zunil.

319 There is a cell near 160° east, 0° north that often contains the maximum
 320 number of overlaps, although with increased cone size it becomes less of a
 321 driver. When the cone has a starting diameter of 50 kilometers, there are
 322 several areas of cells containing the maximum overlaps with most near Zunil.
 323 When the cone is increased to 100 kilometers in diameter, the maximum cells
 324 consolidate considerably and most of the outlying cells are eliminated.

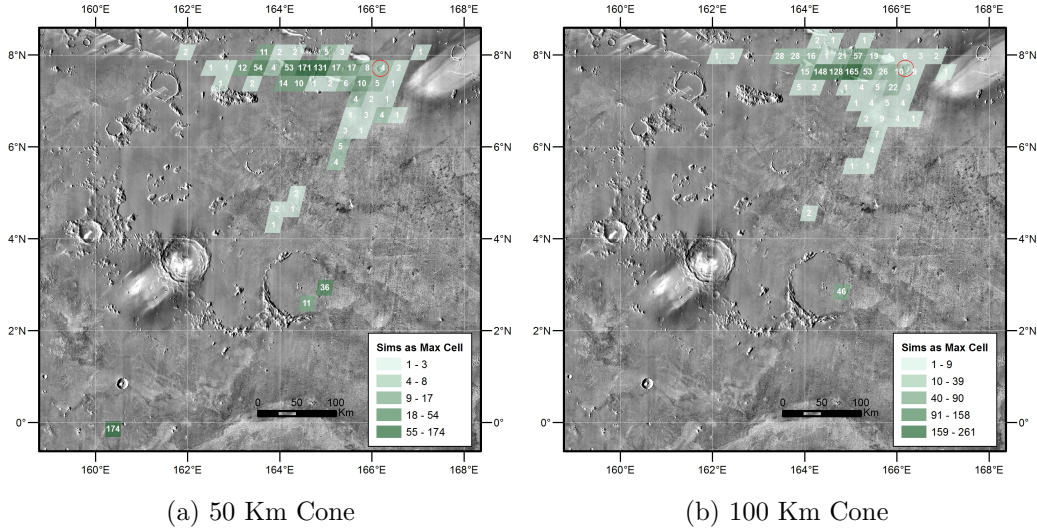


Figure 8: Cells having the maximum count of overlapping cones in at least one of 500 simulations that induce error. Count on each cell represents the number of simulations where it was the maximum.

325 Without prior knowledge of Zunil, it is reasonable to assume that a user
 326 could identify it as the source crater for the secondaries in the test dataset.
 327 In Figure 9 the max cell per test iteration using 100 kilometer cones is shown
 328 with the crater database created by Robbins and Hynek (2012) which con-
 329 tains the location of nearly 400,000 craters having a diameter of one kilometer
 330 or greater. Only three craters from the Robbins and Hynek (2012) dataset
 331 are within a maximum cell, with Zunil being one of them. However, Zunil
 332 falls within a cell that is the maximum in only ten simulations. It is more
 333 likely that for a given iteration Zunil would not fall within the maximum cell
 334 but the trend of the grid cells intersection counts, as described in Section 2.3,
 335 would lead to an investigation of neighborhood craters. With research such
 336 as McEwen et al. (2003) and Preblich et al. (2007), identification of Zunil is

337 possible.

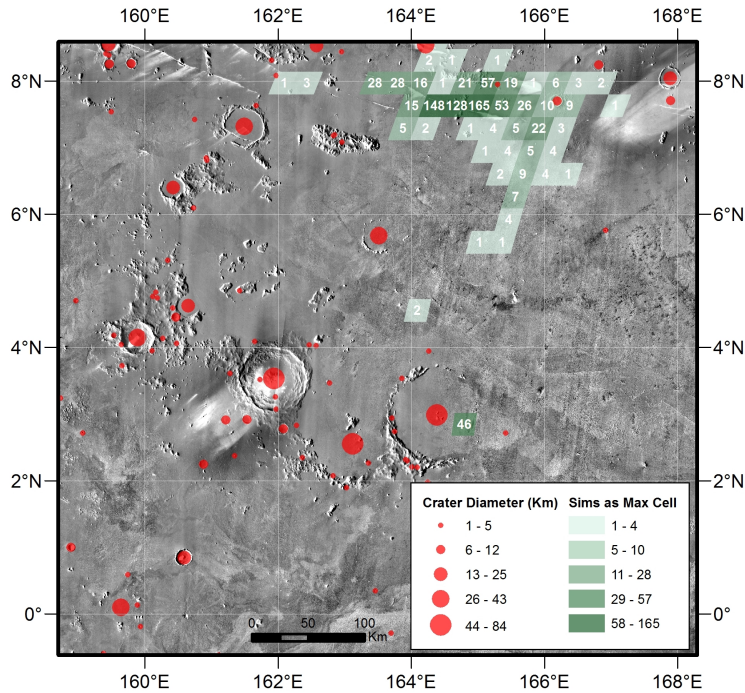


Figure 9: Maximum cells from the 100 kilometer cone error simulation shown with craters having a diameter over one kilometer.

338 3.3. Implications

339 This simulation represents a first phase, best case scenario. Nearly all of
340 the trajectories are based on craters known to be Zunil secondaries, so there
341 is a strong bias of the trajectories to merge toward Zunil. McEwen et al.
342 (2003) showed that Zunil secondaries occur in tight clusters within rays and
343 thus it is likely that the amount of error here is lower than in other systems.
344 This should not invalidate the ability of the cone to compensate for error.
345 Cone size needs to reflect a number of assumptions which will be suited to
346 the crater system being studied. Here we show that the cone of uncertainty

347 can lead to a consistent result. It should not provide pass to ignore accuracy
348 requirements but provide insurance that the end result is correct.

349 **4. Case Studies**

350 As stated previously, the ability of using shapes to effectively identify
351 source craters is a key measurement in assessing the effectiveness of this
352 representation. In addition to the Zunil testing discussed in Sections 2 and
353 3, three case studies were used to examine whether the cone of uncertainty
354 can locate known source craters. In each case the LCC tool was used to create
355 polyline back-projections, then a cone having a diameter of 50 kilometers at
356 the trajectory starting point was created and the subsequent overlaps were
357 enumerated over a grid of 400 square kilometer cells. Each case study below
358 is accompanied by a map of these overlaps, with the darker shading indicating
359 a higher number of intersections.

360 Two of these case studies come from areas on Mars, Mare Acidalium and
361 Lunae Palus, which were used by Laura et al. (2017). This provides a means
362 of comparing the visual interpretation with and without the cone of uncer-
363 tainty. The digitization of craters used for these regions was based on CTX
364 (Malin et al., 2007) and THEMIS (Christensen et al., 2004) imagery having
365 a spatial resolution of eight meters and 100 meters per pixel, respectively.
366 In this paper we describe an additional case study, the entire surface of the
367 moon, as an example from a separate surface and scale. The craters in the
368 lunar dataset was digitized using the Wide Area Camera (WAC) (Robinson
369 et al., 2011) mosaic and a Lunar Orbiter Laser Altimeter (LOLA) (Smith
370 et al., 2010) derived Digital Terrain Model (DTM).

371 *4.1. Acidalia*

372 The Mare Acidalium quadrangle lies in the northeastern plains of the
373 Martian western hemisphere, bounded by 30° to 60° north latitude and 0°
374 to 60° west longitude. There were a total of 44 bi-directional trajectories (88
375 trajectories total) examined in Mare Acidalium based on secondary crater
376 clusters whose bounding ellipse had an inverse flattening less than two. No
377 minimum threshold was placed on the length of the trajectory, in order to
378 keep a robust data set. The resulting cone overlap counts are shown in
379 Figure 10. This data set provides the first opportunity to use the cones of
380 uncertainty in a setting with multiple primary craters.

381 There are two areas in Mare Acidalium that strongly suggest the existence
382 of a primary crater. The first, denoted as A1, is located in the vicinity of
383 29° west, 43° north and the second, denoted as A2, is located in the vicinity
384 of 46° west, 40° north. These areas represent clusters of overlapping cones
385 which intersect at different angles. There are areas which have higher counts
386 as a result of parallel cones in close proximity. The areas northeast and
387 southwest of A1 are such examples that are likely areas of contamination,
388 identified from the pattern in the grid, rather than a string of primaries.
389 There is likely only one or two primaries in the line from 40° west, 30° north
390 to 20° west, 50° north and one is A1. From that line of concentration there
391 is another line connecting it to A2. That intersection is another potential
392 primary. There is likely some contamination, but here the counts are not
393 continuous throughout. There are two concentrated pockets, at roughly 42°
394 west, 38° north and 40° west, 36° north that potentially hold a primary.
395 A final area of interest exists near 14° west, 39° north where the count is

396 not particularly high but is unique compared to the surrounding area and
397 appears it is the result of trajectories from at least two directions.

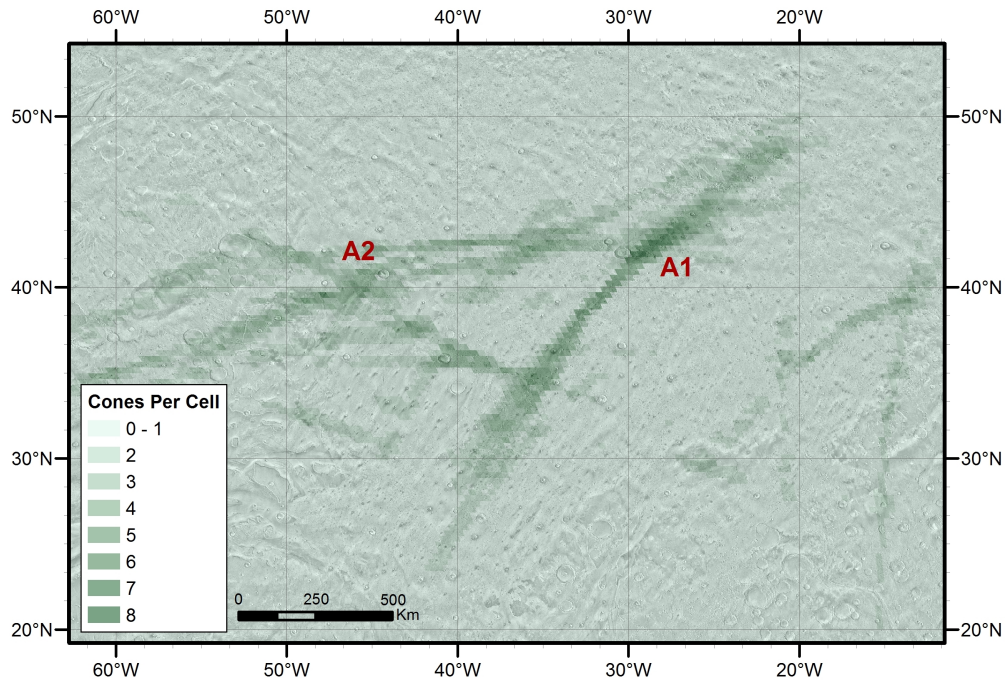


Figure 10: Grid counting overlapping cones of uncertainty in the Mare Acidalium quadrangle, Mars.

398 4.2. *Lunae Palus*

399 The Lunae Palus quadrangle lies in the east-central region of the Martian
400 western hemisphere, bounded by 0° to 30° north latitude and 45° to 90° west
401 longitude. Clusters, catenae, and lineaments were back-projected using the
402 LCC tool for a total of 537 bi-directional trajectories (1,074 single directional
403 trajectories). No restrictions were placed on the features and all craters were
404 back-projected. The resulting cone overlap counts are shown in Figure 11.

405 There are several areas in Lunae Palus that suggest the presence of a
406 primary crater. The first is located near 87° west, 22° north with the highest
407 number of overlaps in the entire region. The second strongest candidate is
408 near 51° west, 3° north. It seems likely that there are other primaries in
409 close proximity. These potentials are near 53° west 11° north, 49° west 8°
410 north, and 57° west 5° north. These could be areas of cone contamination
411 or viable primaries. A final area worth investigating exists near 68° west,
412 18° north. These are the strongest candidates. There are likely many other
413 primaries that exist in Lunae Palus given the number of grid intersections.
414 This data set is unique relative to the other study sites in that the number of
415 trajectories is much higher, so there could be areas with a unique localized
416 count suggesting a primary that is not readily obvious upon visual inspection
417 at a small scale.

418 *4.3. Moon*

419 There were a total of 4,981 secondary crater clusters, catenae, and lineaments
420 from the entire lunar surface that were back-projected using the LCC
421 tool without the Coriolis Effect, then the cones were created and intersected.
422 The resulting map is shown in Figure 12. There are many areas that clearly
423 suggest primary craters, most notably at 20° west, 10° north and at 0° west,
424 36° north. Most interestingly, and unique to this data set, is that there are
425 areas of negative space where very few intersections exist. The most prominent
426 example of this is centered at roughly 92° west, 22° south. Looking at
427 the pattern of intersections around this feature, and directed away from it,
428 suggests that this is an impact basin much larger than any grid cell.

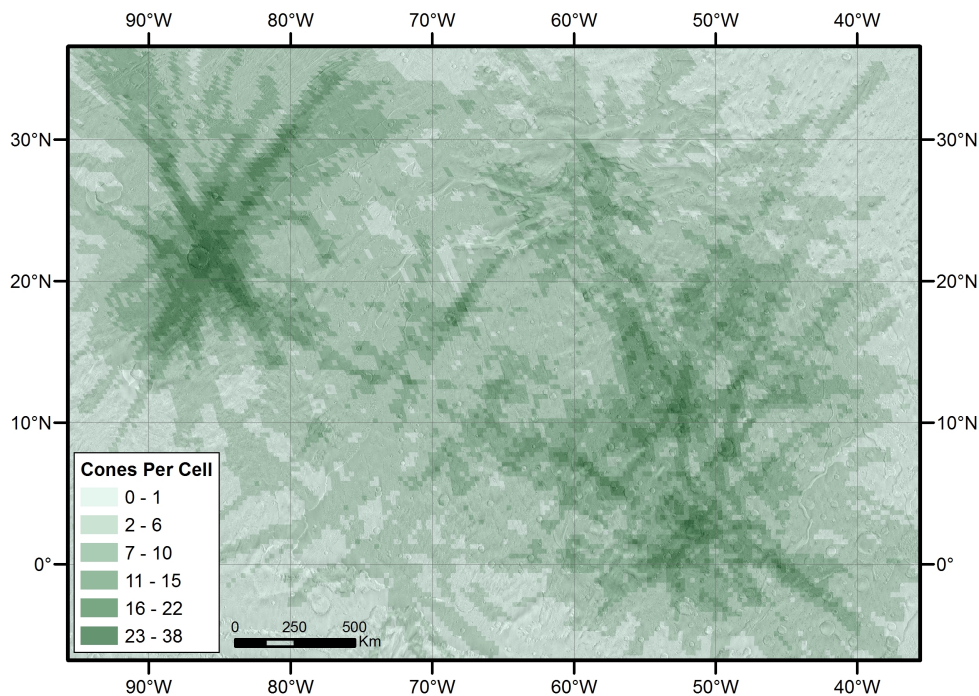


Figure 11: Counts of overlapping cones of uncertainty in the Lunae Palus quadrangle, Mars.

429 *4.4. Commentary on Case Studies*

430 With the exception of the lunar data, the case studies presented here are
 431 common in that they place borders in borderless worlds. The Zunil test case
 432 in particular is biased in that the data are known to be related to Zunil and
 433 few craters independent of Zunil are presented. It may be the case that a
 434 study area with a wider scope makes it more difficult for primaries to be
 435 identified. However, we have seen in the lunar data that in a worldwide data
 436 set primaries are still identifiable. We caution, then, against the selection of
 437 a very refined study area.

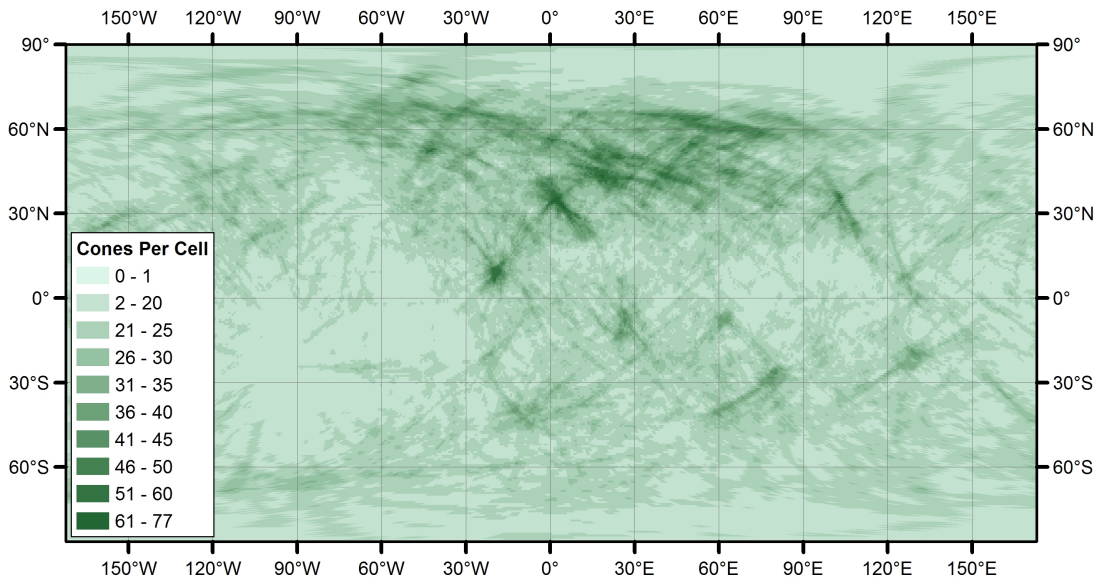


Figure 12: Counts of overlapping cones of uncertainty across the entire lunar surface.

438 **5. Conclusions**

439 In this paper we have established the cone of uncertainty as a meaningful
 440 tool in back-projecting secondary craters to identify source craters. The LCC
 441 tool extension is openly available as a Python script designed to seamlessly
 442 use the output from the LCC tool set. Along with the description of the
 443 construction of the cone the code is open to contribution and refinement.
 444 We have tested this LCC tool extension on three areas of Mars as well as the
 445 moon, and have demonstrated its ability to compensate for trajectory error,
 446 address the MUAP, and quantitatively assess source impact locations.

447 We anticipate the application of using cones during back-projection will
 448 be immediately useful to anyone searching for source craters. It takes us one
 449 step closer in merging the worlds of impact science and crater studies for
 450 geological analysis. These sciences are, naturally, continually developing and

451 should unite to provide a more holistic view of past and current planetary
452 bombardment. Despite this more automated, quantified view there still exists
453 a strong need for human interpretation throughout the process. This paper
454 serves as another tool to support that function.

455 As the LCC tool was a first step in the quantification of back-projecting,
456 so too is the implementation of the cone of uncertainty a first step in ac-
457 counting for trajectory error. We identify four opportunities for future work
458 in this area. First, the quantification of the amount of potential error in the
459 trajectories would be useful quality assurance and in identifying the selection
460 of the cone parameters. Second, more testing around the uncertainty should
461 be undertaken. Each crater system will provide unique challenges and differ-
462 ing amounts of error. Testing multiple systems, and on multiple bodies, will
463 help further refine the cone parameters. Third, the cone parameters should
464 be refined scientifically. This paper analyzed the problem of back-projecting
465 from primarily a spatial point of view, and scientific guidelines should be
466 provided to increase its usability. Finally, whether using polyline or cone
467 back projections, each secondary is being associated with one or more pri-
468 maries. Given the Monte Carlo simulation undertaken herein and the ability
469 to use those simulations to quantify the probability of a secondary to primary
470 association, we intend to explore iteratively removing secondaries from the
471 analysis pool as they are associated with a primary.

472 **6. Acknowledgement**

473 Any use of trade, firm, or product names is for descriptive purposes only
474 and does not imply endorsement by the U.S. government.

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